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Rocket Propulsion Elements: Sutton, George P., Biblarz, Oscar

PROPULSION 3 rd. Eq. 2-7 of Biblarz and Sutton (2001) [1]. De ne some quantities: mass ratio = m f m 0 where m f:= nal mass (after rocket operation had consumed all usable propellant), which is M 0 above m 0:= mass before rocket operation, which is M(0) propellant mass fraction m p M(0) cf. Eq. 2-8 of Biblarz and Sutton (2001) [1]. For thrust F thrust, F thrust = _mu

Contents

Sutton, George Paul. Rocket propulsion elements : an introduction to the engineering of rockets / by George P. Sutton, Oscar Biblarz.--7th ed. p. cm. "A Wiley-Interscience publication." Includes bibliographical references and index. ISBN 0-471-32642-9 (cloth: alk. paper) 1. Rocket engines. I. Biblarz, Oscar. II. Title

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